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(54) Title: **SHAPE MEMORY ALLOY ARTICLES WITH IMPROVED FATIGUE PERFORMANCE AND METHODS THEREFORE**

(57) Abstract: Articles made of shape memory alloys having improved fatigue performance and to methods of treating articles formed from shape memory alloy materials by pre-straining the articles (or desired portions of the articles) in a controlled manner so that the resultant articles exhibit improved fatigue performance. The shape memory articles are preferably medical devices, more preferably implantable medical devices. They are most preferably devices of nitinol shape memory alloy, most particularly that is superelastic at normal body temperature. The pre-straining method of the present invention as performed on such articles includes the controlled introduction of non-recoverable tensile strains greater than about 0.20% at the surface of a desired portion of a shape memory alloy article. Controlled pre-straining operations are performed on the shape-set nitinol metal to achieve non recoverable tensile strain greater than about 0.20% at or near the surface of selected regions in the nitinol metal article. The pre-straining operations result in a significant increase in fatigue life of the selectively treated regions and an overall improvement in the fatigue performance of the device.

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TITLE OF THE INVENTION

SHAPE MEMORY ALLOY ARTICLES WITH IMPROVED FATIGUE PERFORMANCE AND
METHODS THEREFORE

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FIELD OF THE INVENTION

The present invention relates to the field of shape memory alloys, medical articles
made from shape memory alloys and more specifically to shape memory alloy articles
10 having improved fatigue resistance, and methods of making such articles.

BACKGROUND OF THE INVENTION

Shape memory alloys have been used for a variety of applications since the
15 discovery of shape memory transformation by Chang and Reed in 1932. Nitinol, the near-
equiatomic alloy of nickel and titanium (optionally containing ternary, quaternary or more
elements such as copper, chromium, iron, vanadium, cobalt or niobium) that thus far offers
the most useful shape memory and superelastic properties, was discovered by Buehler and
his colleagues in 1962.

20 Nitinol has proven to be adequately biocompatible for a variety of medical device
applications, including implantable applications. It has been used for orthodontics, in a
variety of orthopedic devices, for filter devices in various body conduits and for stent devices
for maintaining patency of various body conduits, particularly those of the vasculature.
These stent devices (including stent-grafts, i.e., stents provided with a flexible covering of a
25 vascular graft material such as porous expanded polytetrafluoroethylene) are generally
inserted into a body conduit at a site remote from the intended implantation location, and
transported to the desired location by a catheter or similar device. They are usually inserted
in a collapsed or compacted state to enable their movement through the body conduit to the
desired implantation site, at which location they are expanded to the desired size to
30 interferably fit within the conduit and hold the conduit open at that location. While these
devices are most often used for cardiac applications, they are also used for the repair of
thoracic and abdominal aortic aneurysms and for peripheral and carotid applications.

Many of these stent devices are made from materials intended to be expanded by
the application of a force applied internal to the tubular device, usually by the inflation of a
35 catheter balloon on which the device was mounted for insertion into the body conduit. These
balloon-expandable devices are most often made from a plastically deformable material

such as a stainless steel. Many other stents are made from shape memory materials, particularly nitinol, and take advantage of the shape memory or superelastic properties so that they may be implanted simply by releasing the constrained, compacted device and allowing it to self-expand at the desired implantation site.

5 Stent devices should be adequately flexible to enable them to be delivered through bends in the sometimes-tortuous pathways of a body conduit. They may also need to be adequately flexible to conform to bends in the body conduit at the implantation site, and to be able to accommodate movement of the body conduit. This is particularly true in the vasculature, where a vessel often changes dimension as a function of systole and diastole.
10 These devices consequently should also have good fatigue resistance.

Shape memory materials can exhibit pseudoelastic (superelastic) behavior, allowing the material to recover a significant amount of strain due to the reversible, isothermal metallurgical phase transformations by changes in the state of stress. The superelastic behavior is characterized by a linear elastic and a nonlinear pseudoelastic stress-strain
15 response allowing the material to recover a significant amount of strain due to the reversible austenitic-martensitic phase transformation. Conventional nitinol materials can typically recover principle strains on the order of up to 8% (see "Nitinol Medical Device Design Considerations" by Philippe P. Poncet, SMST-2000: Proceedings of the International Conference on Shape Memory and Superelastic Technologies, pp. 441-455). The
20 superelastic behavior of nitinol allows for the design of devices that exert a relatively constant stress over a wide range of strains or shapes. This unique behavior has been utilized in the design of many implantable medical devices such as stents and stent-grafts.

The phase stability of nitinol is a function of both temperature and stress. The phase stability in the unstressed state is characterized by the transformation temperatures M_s , M_f , A_s , and A_f . Martensite is the stable phase at temperatures below M_f , the martensitic finish temperature. Upon heating, the martensitic structure begins a reversible thermoelastic phase transformation to austenite when the temperature reaches A_s , the austenitic start temperature. The transformation to austenite is completed when the temperature reaches A_f , the austenitic finish temperature. Upon cooling the austenite, the material begins to
30 transform to martensite at a temperature equal to M_s , the martensitic start temperature, and completes its transformation to martensite at a temperature equal to M_f , the martensitic finish temperature.

The shape memory effect of nitinol is demonstrated by shaping the material in the relatively high-temperature austenitic phase and setting the shape by an appropriate heat
35 treatment. Upon cooling the material below the martensitic transformation temperature, the material can be deformed to a second shape configuration while in the martensitic state.

Upon heating to temperatures above the austenitic transformation temperature the material will return to its original shape configuration. Conventional nitinol materials can typically recover up to 8% strain by this shape memory effect (reference ASM Handbook, Volume 2, Shape Memory Alloys, Darel Hodgson et al., page 899).

5 The superelastic effect of nitinol is demonstrated by the application of stress to the nitinol material at temperatures above the austenitic transformation temperature, and below M_d , the maximum temperature at which stress-induced martensite can be formed. The initial application of stress in this case causes the austenitic structure to deform in the classical Hookean linear elastic manner until a critical stress is achieved. The application of stress
10 beyond this critical stress results in a nonlinear stress-strain response due to the isothermal reversible transformation to martensite. Upon removal of the applied stress, the material can reversibly transform back to austenite, returning to its original shape. As noted previously, conventional nitinol materials can recover approximately 6-8% strain by this superelastic effect.

15 The alternating in-vivo load conditions (due to changes such as between systole and diastole) often limit the design of medical devices such as stents and stent-grafts due to the fatigue capability of nitinol materials. Improvements in the fatigue performance of nitinol are desirable to provide an increased fatigue life and fatigue life safety factor and to increase design flexibility for implantable medical devices that include nitinol.

20 Various publications describe the fatigue resistance of devices made from shape memory materials. European Patent Application EP1170393 describes a method for improving fatigue performance of actuators made from materials that have shape memory effect. The process includes introducing significant cold work, applying stress in the expected loading direction, and heating above the recrystallization temperature for short
25 times to create a uniform, fine-grained, microstructure.

 According to a published article, "Cyclic Properties of Superelastic Nitinol: Design Implications" (SMST-2000: Proceedings of the International Conference on Shape Memory and Superelastic Technologies, D. Tolomeo, S. Davidson, and M. Santinoranont, pp. 471-476) strain-controlled fatigue tests were conducted with various pre-strain conditions up to
30 6% pre-strain. Samples were subjected to strains up to 6%, then unloaded to a specified cyclic displacement. The endurance limits for different pre-strain values remained relatively constant.

 A published article titled "Effect of Constraining Temperature on the Postdeployment Parameters of Self-Expanding Nitinol Stents" (SMST-2000: Proceedings of the International
35 Conference on Shape Memory and Superelastic Technologies, Martynov and Basin, pp. 649-655) describes the evaluation of retaining temperature on the post deployment

parameters of 28mm aortic-size stents having a typical diamond shaped stent cell structure. The article states that "The maximum deformation of any stent element in the fully compressed state (when the stent is packed into a delivery catheter) should not exceed the available reversible deformation limit, which is about 6 to 8%, depending on the material used."

In another published article, "Fatigue and Fracture Behavior of Nickel-Titanium Shape Memory Alloy Reinforced Aluminum Composites," authors Porter and Liaw describe an aluminum matrix composite reinforced with discontinuous nitinol particulates by powder metallurgy processing. The reinforced composite material is cold rolled at minus thirty degrees centigrade (-30°C). Upon re-heating, the nitinol transforms to austenite creating residual internal stresses around each particle to strengthen the material. Improved fatigue life were observed compared to the unreinforced control matrix material.

An article entitled "The Study of Nitinol Bending Fatigue" (W.J. Harrison and Z.C. Lin, SMST-2000, Proceedings of the International Conference on Shape Memory and Superelastic Technologies) describes fatigue testing of nitinol samples subjected to alternating strain to simulate the effects of changing strain resulting from systole and diastole, and optionally subjected to an additional constant strain (mean strain) that would be expected to result from the interference between an expanded stent and the vessel into which it has been fitted. The samples tested were cut from nitinol tubing. The samples showed good fatigue life, with the fatigue life being greater for samples exposed to higher mean strain. This result suggests that the samples had apparently been cut at their small diameter (i.e., the "compacted" diameter appropriate for insertion of such a device into a body conduit) and subsequently expanded to a larger diameter at which they were tested, as opposed to having been cut at the larger, expanded diameter and then compressed slightly to create the mean strain.

SUMMARY OF THE INVENTION

The present invention relates to articles made of shape memory alloys having improved fatigue performance and to methods of treating articles formed from shape memory alloy materials by pre-straining the articles (or desired portions of the articles) in a controlled manner so that the resultant articles exhibit improved fatigue performance.

The shape memory articles are preferably medical devices, more preferably implantable medical devices. They are most preferably devices of nitinol shape memory alloy, most particularly that is superelastic at normal body temperature (approximately 37°C).

Implantable medical devices are those devices that are intended to remain within a living body for periods of 24 hours or longer.

The shape memory alloy articles may be produced from materials of various shapes, such as wire of various transverse cross sectional shapes including circular, elliptical, square, rectangular, etc. Alternatively, the articles may be made by machining precursor forms such as sheets, tubes or rods, as by electrical discharge machining (EDM), laser cutting, chemical milling, or the like.

The pre-straining method of the present invention as performed on such articles includes the controlled introduction of non-recoverable tensile strains greater than about 0.20% at the surface of a desired portion of a shape memory alloy article. Controlled pre-straining operations of the shape-set nitinol metal are performed to achieve non-recoverable tensile strain greater than about 0.20% at or near the surface of selected regions in the nitinol metal article. The pre-straining operations result in a significant increase in fatigue life of the selectively treated regions and an overall improvement in the fatigue performance of the device. The pre-straining treatments described in this invention are useful for increasing the fatigue life safety factor of current nitinol-based medical devices and for incorporating into the design of future implantable medical devices that include nitinol, thereby providing additional design flexibility.

Controlling the amount of pre-strain involves pre-straining the shape memory metal by the controlled application of bending, torsional or a combination of these and/or other forces at pre-determined temperatures. These amounts of pre-strain (resulting in at least about 0.20% non-recoverable strain) may be calculated by analytical methods such as finite element analysis or the like, in conjunction with the material's loading and unloading behavior.

Non-recoverable tensile strain is intended to mean the permanent set, i.e., the plastic deformation that remains upon releasing the tensile pre-strain or stress, arising from the displacement of atoms to new lattice sites, as determined by representative material stress-strain (loading and unloading) behavioral properties, or as measured by techniques such as microhardness testing, x-ray diffraction, backscatter electron Kikuchi patterns, synchrotron radiation, convergent beam electron diffraction or the like.

The method of this invention involves pre-straining articles such that targeted surface regions are subjected to tensile pre-strains exceeding the recoverable strain limit of the material (typically 6%-8% strain), while maintaining a significant portion of the subsurface area (less affected by the pre-strain) within the superelastic material limit. Tensile pre-strains of this type may be induced by the application of forces such as bending or torsional forces. Upon removing the pre-straining force, the lesser-affected superelastic subsurface

region of the article allows the bulk article to recover a significant level of strain, such that the article, following the removal of the pre-straining force, returns to or near to its original geometry.

This process thus results in desired local surface regions of the pre-strained article being in a state of compression. A residual compressive stress state has thus been induced at the targeted surface region. The result is a significant improvement in fatigue performance of targeted regions of the article subjected to this pre-straining operation due to the introduction of residual compressive surface stresses.

The process of inducing compressive residual surface stresses at desired locations by the controlled pre-straining operation of the present invention, may also produce a concomitant surface region which is subjected to compression, on the side of the article opposite the targeted region subjected to tension during the pre-straining operation. The compressive strains introduced on the regions opposite the targeted regions may also exceed the recoverable strain limit of the material, resulting in a residual state of tension at these regions upon removal of the pre-straining load. The end result of the pre-straining operation disclosed in this invention is the improvement in fatigue performance at the targeted regions of the medical article, thus resulting in a more fatigue resistant device. This operation can thus be applied to specifically chosen regions of a medical device where service fatigue loading is most severe and improved fatigue performance is desired, or over the entire surface region of the article.

BRIEF DESCRIPTION OF THE DRAWINGS

Figure 1A shows a perspective side view of a nitinol alloy wire of circular cross-section subjected to a pre-straining operation by a controlled bending operation.

Figure 1B shows a transverse cross-sectional view taken through the wire of Figure 1A indicating representative strain contours for the selectively treated cross-sectional area of the wire.

Figure 1C shows a view of a shape-set nitinol wire specimen: test specimens are pre-strained following the shape-set heat treatment, while control specimens are not.

Figure 2 shows stress-strain curves for nitinol wires subjected to pre-straining that results in non-recoverable tensile strain of less than about 0.20% and for inventive wires subjected to tensile pre-straining at 37°C that results in non-recoverable tensile strain of greater than about 0.20%.

Figure 3 shows stress-strain curves for nitinol wires subjected to pre-straining that results in non-recoverable tensile strain of less than about 0.20% and for inventive wires that results in non-recoverable tensile strain of greater than about 0.20%, loaded at -30°C in tension followed by unloading at -30°C from various pre-strain levels, and heated in the stress-free state to 37°C .

Figure 4 shows stress-strain curves for nitinol wires loaded at -30°C in tension, heated to 37°C while maintained at various pre-strain levels, followed by unloading at 37°C from the various pre-strain levels.

Figure 5 shows a graph of the non-recoverable strain achieved in nitinol wire when subjected to various methods of tensile pre-straining.

Figure 6 shows stress-strain curves for nitinol wires subjected to tensile pre-straining treatments at various elevated temperatures.

Figure 7 shows a fitted Weibull fatigue survivability plot for a group of nitinol wire samples provided with tensile pre-straining treatment in accordance with embodiments of the present invention, compared to a non pre-strained control group when both groups were subjected to an axial fatigue test.

Figure 8 shows a fitted Weibull fatigue survivability plot for a group of nitinol wire samples provided with bending pre-straining treatment in accordance with embodiments of the present invention, compared to the non pre-strained control group when both groups were subjected to a flexural fatigue test.

DETAILED DESCRIPTION OF THE INVENTION

The present invention relates to methods of treating implantable medical device components formed from nitinol materials (such as nitinol wire) so that the resultant device exhibits improved fatigue performance. This invention identifies methods to induce non-recoverable tensile strain greater than about 0.20% strain at or near the surface of selected regions of the nitinol metal by controlled pre-straining processes. The non-recoverable tensile strain may thus be greater than about 0.25%, 0.3%, 0.35%, 0.4%, 0.45%, 0.5%, 0.6%, 0.7%, 0.8%, 1.0%, 1.25%, 1.5%, and 2.0%.

Implantable medical devices are typically designed such that the maximum deformation of any portion of the nitinol material does not exceed the available reversible deformation limit, typically 6% to 8% strain, after the shape setting treatments ("Effect of Constraining Temperature on the Postdeployment Parameters of Self-Expanding Nitinol Stents," SMST-2000: Proceedings of the International Conference on Shape Memory and

Superelastic Technologies, Martynov and Basin). The purpose of maintaining maximum principle strains below the reversible deformation limit after the shape setting process has been completed is to ensure that the device will preserve its original shape.

The reversible deformation limit is defined as the maximum strain a material can undergo without inducing non-recoverable strain (i.e., permanent set) greater than about 0.20%.

The methods of this invention involve the controlled pre-straining of desired portions of nitinol articles such that targeted surface regions are subjected to tensile strains exceeding the about 0.20% recoverable strain limit of the material, while maintaining a significant portion of the cross-section within the superelastic material limit. Upon removing the pre-straining force, the superelastic region of the structure allows the bulk structure to recover significant levels of strain such that the structure returns to, or near to, its original geometry. This process thus results in desired regions of the pre-strained material that had been subjected to tensile strains beyond their recoverable limit to be in a state of compression. A residual compressive stress state has thus been induced at the targeted surface regions. This process results in a significant improvement in fatigue performance of targeted regions of the structure subjected to this pre-straining operation due to the introduction of residual compressive surface stresses. The controlled process of pre-straining can be accomplished by flexural loading, torsional loading, or any combination of loading conditions designed to induce non-recoverable tensile strains greater than about 0.20% at or near the surface of fatigue-critical locations of a shape-set, nitinol containing implantable medical device.

An example of a pre-straining method included as an embodiment of this invention is illustrated in Figures 1A and 1B. Figure 1A shows a nitinol wire of circular cross-section being deformed by a controlled bending pre-straining operation, indicated by the arrows. The transverse cross-section of Figure 1B shows constant strain contours of the indicated cross-section of the wire during the pre-straining, bending operation in accordance with Figure 1A, with the region showing arcuate iso-strain contours near location A representing the region subjected to non-recoverable tensile strains greater than about 0.20%. In Figure 1B, the region showing the iso-strain contours near location B is subjected to compressive strains. Upon removal of the pre-straining force, the superelastic nature of the bulk material (i.e., generally the material outside of the regions indicated by the iso-strain contours) forces the material to return to, or near to, its original shape. This operation thus induces residual compressive stresses at or near the surface location A and residual tensile stresses at or near the surface location B. The result of this pre-straining operation is an improvement in

the fatigue performance of the treated region A. This pre-straining method may thus be applied in a controlled manner to treat desired fatigue-critical locations of a device.

Figure 1C shows a view of a shape-set nitinol wire specimen. Wire specimen 10 is formed around pins 11, 12, and 13 and includes about 1.25 turns of the wire around each of pins 11 and 13 as shown, to create loops 14. The two opposing loops (14) are wound in opposite directions (i.e., clockwise and counter-clockwise). The distance between pins 11 and 12 is defined as dimension "A" with dimension "B" (partially defining the location of pin 12, parallel to dimension "A") being half of dimension "A". Dimension "C" finally defines the location of pin 12 as the distance pin 12 is located above a line between pins 11 and 13. Pins 11 and 13 are of equal diameter. The diameter of pin 12 is chosen to provide the desired radius at the apex 15 of the wire specimen. After being formed as shown, these wire specimens 10 are subjected to shape-set heat treatment prior to fatigue testing. Test specimens are pre-strained following the shape-set heat treatment, while control specimens are not.

Figure 2 shows stress-strain curves for nitinol wire samples loaded at 37°C in tension, followed by unloading at 37°C, from various pre-strain levels. It is noted that the particular stress-strain response is dependent upon such factors as alloy composition and thermal and mechanical process histories. The resultant non-recoverable tensile strain (i.e., permanent set) increases with increasing pre-strain level. Information obtained from this type of family of stress-strain curves, in conjunction with analytical procedures such as finite element analysis, can be utilized to develop an appropriate temperature, controlled pre-straining (bending) process. This process is designed to induce non-recoverable tensile strain levels greater than about 0.20% at desired, fatigue-critical surface locations of a shape-set nitinol structure. It is apparent this process can be developed for other temperatures as well.

Figure 3 shows a representative family of stress-strain curves for nitinol wire samples which have been loaded at -30°C in tension, followed by unloading at -30°C from various pre-strain levels, and heated in the stress-free state to 37°C. The resultant non-recoverable tensile strain increases with increasing pre-strain level. This family of stress-strain curves, in conjunction with analytical procedures such as finite element analysis, can similarly be utilized to develop an appropriate temperature, controlled pre-straining process designed to induce non-recoverable tensile strain levels greater than about 0.20% at fatigue-critical surface locations of a shape-set nitinol structure.

Figure 4 shows another representative family of stress-strain curves for nitinol wire samples loaded to various pre-strain levels at -30°C in tension, heated to 37°C while maintained at their respective pre-strain condition, followed by unloading at 37°C from their

respective pre-strain condition. The resultant non-recoverable tensile strain increases with increasing pre-strain level. This family of stress-strain curves, in conjunction with analytical procedures such as finite element analysis, can similarly be utilized to develop an appropriate temperature, controlled pre-straining process designed to induce non-recoverable tensile strain levels greater than about 0.20% at fatigue-critical surface locations of a shape-set nitinol structure.

Figure 5 shows a plot of non-recoverable tensile strain as a function of tensile pre-strain level for the various controlled pre-straining procedures described in Figures 2-4. Curve A describes samples subjected to tensile pre-strain at 37°C and then unloaded at 37°C (as shown in Figure 2). Curve B describes samples subjected to tensile pre-strain at -30°C, unloaded at -30°C and then heated to 37°C (as shown in Figure 3). Curve C describes samples subjected to pre-strain at -30°C and then heated to 37°C in the pre-strained condition, and subsequently unloaded at 37°C (as shown in Figure 4). This type of plot, in conjunction with analytical procedures such as finite element analysis, can be utilized to develop an appropriate temperature, controlled pre-straining process designed to induce non-recoverable tensile strain levels greater than about 0.20% at fatigue-critical locations of a shape-set nitinol structure.

Additional techniques may be utilized to decrease the recoverable strain limit for given pre-strain levels to allow for the introduction of the desired non-recoverable tensile strain at the fatigue-critical locations. These techniques can be incorporated to allow for the introduction of non-recoverable tensile strains at relatively low pre-strain levels (less than 6% to 8% pre-strain). Such techniques include but are not limited to, chemical compositional alloy modifications, thermal and mechanical process history modifications, surface modification techniques such as laser surface treatments, or the like.

Figure 6 shows a family of stress-strain curves for nitinol wire samples loaded in tension to 6% pre-strain and unloaded at various temperatures. The resultant non-recoverable tensile strain is shown to increase with increasing temperature. This provides an example of one technique, by using an elevated temperature pre-straining, which can be used to create relatively high (greater than about 0.20%) non-recoverable strains at relatively low pre-strain levels. This type of plot can be used, in conjunction with analytical procedures such as finite element analysis, to develop an appropriate temperature, controlled pre-straining process designed to induce non-recoverable tensile strain levels greater than about 0.20% at fatigue-critical locations of a shape-set nitinol structure. The use of elevated temperature, controlled pre-straining processes can be utilized to induce non-recoverable tensile strain levels greater than about 0.20% at fatigue-critical locations of a shape-set nitinol structure.

Another technique to provide for the induction of significant (greater than about 0.20%) non-recoverable tensile strains at relatively low pre-strain levels includes the use of a composite structure consisting of a superelastic nitinol core material and an outer surface material with limited recoverable strain capability. The outer material may include a nitinol material with an A_f greater than 37°C, preferably a nitinol material with an A_s greater than 37°C. Alternatively, the outer surface material may also be stainless steel, or any other material with a lower recoverable strain limit than the nitinol core material. The use of such a composite material can allow the introduction of significant non-recoverable tensile surface strains at relatively low pre-strain levels. The induction of non-recoverable tensile surface strains greater than about 0.20% at or near the surface may be introduced by pre-straining the material by bending pre-straining, torsional pre-straining, or a combination of complex pre-strain loading conditions.

The process of inducing compressive residual surface stresses by the pre-straining operations described herein may also produce a concomitant surface region which is subjected to compressive strains, occurring on the opposite surface region of the targeted region subjected to tension, during the pre-straining operation. The compressive strains introduced on the regions opposite the targeted regions may also exceed the recoverable strain limit of the material, resulting in an undesirable residual state of tension at these regions which may result in reduced fatigue life.

Figure 1C shows the test specimen in a relaxed condition, wherein apex 15 contains no significant residual stresses. Following controlled pre-straining caused by moving pins 11 and 13 closer together, the outer radius of apex 15 of the specimen as shown in Figure 1C will be in the state of residual compression stress while the inner radius will be in the state of residual tensile stress. This method of pre-straining is thus desired when the critical fatigue location is the outer radius of apex 15. Alternatively, if the fatigue-critical location is the inner radius of apex 15, pre-straining is accomplished by moving pins 11 and 13 further apart. In service, the fatigue-critical location is one that has been previously pre-strained in tension, thus inducing residual compressive stress at that fatigue-critical location.

The end result of the pre-straining operation disclosed in this invention is the improvement in fatigue performance at targeted regions of the medical device structure, thus resulting in a more fatigue resistant device. This operation can thus be applied to specific medical device structure regions where service fatigue loading is most severe and improved fatigue performance is desired, or over the entire surface region of the structure.

In another aspect of the present invention, it is noted that it is not uncommon for nitinol articles including implantable medical articles to be subjected to surface modification by various methods such as electropolishing and shot peening. These methods are known

to reduce any non-recoverable strain at the surface of these articles. Consequently, it is appropriate that any desired surface modification is performed prior to the controlled pre-straining operations as taught by the method of the present invention.

5 Example 1:

Axial fatigue tests were conducted using superelastic nitinol wire samples subjected to different tensile pre-strain conditions. The nitinol wire (Fort Wayne Metals, Fort Wayne, IN, nominal diameter 0.305mm) utilized for these tests was electropolished to a diameter of 0.300mm and heat treated in air to obtain a straight configuration and to impart superelastic
10 behavior at 37°C ($A_f < 37^\circ\text{C}$) with a permanent set of less than 0.20% when loaded to 6% strain and unloaded at 37°C.

An Instron servohydraulic test machine (Canton, MA, model no. 8841) was used for the axial fatigue testing. The testing was performed in an air thermal chamber set at 37°C (+/- 1°C). Wavemaker software (Fast Track 2, Wavemaker Editor/Runtime, version 7.0.0,
15 provided by Instron) was used to generate and execute the axial fatigue tests using a displacement controlled sine waveform. Test specimen gauge length was 100mm, held with flat-faced grips (Instron PN 2716-016). Five specimens were pulled to 104mm length (4% mean strain), and cycled $\pm 0.5\text{mm}$ (0.5% alternating strain) at cyclic frequencies until failure by fracture, as shown in Table 1. Three additional specimens were pulled to 108mm length
20 (8% pre-strain), released to 104mm length (4% mean strain), and cycled $\pm 0.5\text{mm}$ (0.5% alternating strain), at a cyclic frequency of 12 Hz until fracture (Table 2). An additional three specimens were pulled to 106mm length (6% pre-strain), released to 104mm length (4% mean strain), and cycled $\pm 0.5\text{mm}$ (0.5% alternating strain), at a cyclic frequency of 12 Hz until fracture (Table 3).

25 Test results as presented in Tables 1-3 show an increase in fatigue life with an increase in pre-strain level.

TABLE 1: No pre-strain, 4% mean strain, 0.5% alternating strain

Specimen	cyclic frequency	CTF
1	8 Hz	3,852
2	8 Hz	2,998
3	15 Hz	3,383
4	12 Hz	3,868
5	12 Hz	3,988

mean CTF: 3618 cycles

30

TABLE 2: 8% pre-strain, 4% mean strain, 0.5% alternating strain

Specimen	cyclic frequency	CTF
6	12 Hz	9,266
7	12 Hz	9,779
8	12 Hz	9,533

mean CTF: 9526 cycles

TABLE 3: 6% pre-strain, 4% mean strain, 0.5% alternating strain

Specimen	cyclic frequency	CTF
9	12 Hz	6,185
10	12 Hz	7,520
11	12 Hz	7,541

mean CTF: 7082 cycles

5 The axial fatigue test results are summarized in Figure 7, showing a fitted Weibull distribution fatigue survival plot comparing the fatigue lives for different groups of nitinol wire samples (plotted as proportion of survivors within each group versus number of cycles to failure, or CTF).

10 Example 2:

Flexural fatigue tests were conducted using superelastic nitinol wire (Fort Wayne Metals, Fort Wayne, IN, nominal diameter 0.323mm) samples subjected to different tensile pre-strain conditions. The nitinol wire used for these tests was electropolished to a diameter of 0.321mm.

15 Thirty wire test specimens were formed into the shape described in Figure 1C, by winding the wire around the 0.79mm diameter stainless steel pins 11, 12 and 13 of the heat treatment fixture, as shown in Figure 1C. All test specimens were heat treated in air while on the fixture to set the test sample geometry configuration and to impart superelastic behavior at 37°C ($A_f < 37^\circ\text{C}$) with a permanent set of less than 0.20% when loaded to 6% strain and unloaded at 37°C. Dimension "A" between pins 11 and 13 (center-to-center) was 20 13.72mm, while dimension "B" was half of dimension "A". Dimension "C" was 5.08mm. The support loops 14 at the ends of each sample 10 were of an inside diameter that conforms to the diameter of pins 11 and 13. The apex 15 of each test specimen 10 was formed to a radius (at the inside radius of the apex bend) that conformed to the diameter of pin 12.

25 Prepared test specimens were divided into three separate groups (10 samples per group): a control group (Group 1: no pre-strain), a room temperature pre-strain group (Group 2), and a cold pre-strain group (Group 3). Each sample from Group 2 was pre-strained by placing the sample eyelet support loops (14) onto the same pin (11) to pre-strain the test sample apex (15) at room temperature. The test specimens were kept at this condition for 2 30 hours at room temperature and then removed. Group 3 samples were placed into a bath mixture of dry ice and 100% isopropyl alcohol, with a submersed thermocouple to monitor bath temperature. The samples were then pre-strained while submersed in the bath, following the same pre-strain procedure described for Group 2. The bath temperature ranged from -34°C to -14°C during the pre-straining procedure. The samples were removed 35 from the bath while in the constrained condition and placed in room temperature air for 2

hours prior to removal of the pre-strain constraint. Group 1 was not pre-strained, and served as control samples for the subsequent fatigue tests. The maximum principle tensile pre-strain level at the fatigue-critical location (outside radius surface of the apex) was calculated to be approximately 8.5%. This maximum principle pre-strain level was calculated by
5 applying standard engineering mechanics formulas (straight and curved beam deflection equations, from "Roark's Formulas for Stress & Strain," 6th edition, McGraw Hill, New York, NY) to the specimen geometry.

Fatigue tests were conducted using a fatigue tester designed and built for the purpose of conducting cyclic, deflection controlled, fatigue testing of apical wire samples of
10 the previously described geometry. The tester is designed to accommodate up to forty test samples. Wire fatigue test samples were loaded onto the fatigue test apparatus by placing the test sample support loops onto 0.79mm diameter stainless steel pins of the fatigue tester. The tester was set to alternate test pin deflections from 9.20mm and 10.16mm (i.e., dimension "A" of Figure 1C alternated between 9.20mm and 10.16mm). These deflections
15 were selected to achieve a maximum principle mean tensile strain of 2.5% and an alternating strain of 0.3% at the outside radius of the test specimen apex. The maximum principle strains for these deflections was calculated by applying standard engineering mechanics formulas (straight and curved beam deflection equations, from "Roark's Formulas for Stress & Strain," 6th edition, McGraw Hill, New York, NY) to the specimen geometry.

20 These deflections were set-up using a telescoping dial depth gage and gage blocks. All 30 specimens were mounted on the tester, with test and control samples being placed alternately along the test fixture. The fatigue testing was performed in a $37\pm1^{\circ}\text{C}$ water bath and at a cyclic frequency of approximately 18 Hz.

The flexural fatigue test results are summarized in Figure 8, showing a fitted Weibull
25 distribution fatigue survival plot comparing the fatigue lives for different groups of nitinol wire samples (plotted as proportion of survivors within each group versus number of cycles to failure by fracture at the apex, or CTF). Data are presented in Tables 4-6 for Groups 1-3 respectively. Various specimens from Groups 2 and 3 survived the 40 million cycle length of the tests as noted. The fatigue test results demonstrate an improvement of approximately
30 three orders of magnitude in the mean fatigue lives for the pre-strained sample groups. This example demonstrates the significant improvement in the fatigue performance of nitinol (particularly nitinol wire) when subjected to a pre-straining treatment.

TABLE 4: Controls, No Pre-Strain

Specimen	CTF
1	13,002
2	17,004
3	20,000
4	20,000
5	23,006
6	24,002
7	24,002
8	24,002
9	29,006
10	37,002

TABLE 5: Room Temperature Pre-Strain

Specimen	CTF
1	125,055
2	1,300,000
3	4,148,832
4	4,246,188
5	12,408,376
6	40,000,000+
7	40,000,000+
8	40,000,000+
9	40,000,000+
10	40,000,000+

TABLE 6: Cold Pre-Strain

Specimen	CTF
1	755,022
2	2,229,536
3	2,399,999
4	2,481,166
5	2,817,037
6	7,723,746
7	8,242,257
8	9,278,477
9	40,000,000+
10	40,000,000+

While the principles of the invention have been made clear in the illustrative embodiments set forth herein, it will be obvious to those skilled in the art to make various modifications to the structure, arrangement, proportion, elements, materials and components used in the practice of the invention. To the extent that these various modifications do not depart from the spirit and scope of the appended claims, they are intended to be encompassed therein.

We claim:

1. A method for producing a nitinol implantable medical device with greater fatigue resistance comprising:
 - a) providing an implantable medical device comprised of nitinol metal exhibiting superelastic behavior at body temperature; and
 - b) pre-straining the nitinol metal in a controlled manner to an extent that selectively induces at least about 0.20% non-recoverable tensile strain at or near the surface of at least one selected location on the nitinol implantable medical device.
2. A method according to claim 1 wherein the implantable medical device is pre-strained at a specific temperature and subsequently exposed to other temperatures in a constrained condition thereby inducing the non-recoverable strain to at least about 0.20% at or near the surface of the at least one selected locations on the nitinol implantable medical device.
3. A method according to claim 1 wherein the implantable medical device comprises a composite material having a superelastic nitinol metal core and an outer surface material with limited recoverable strain capability.
4. A method according to claim 3 wherein the outer surface material with limited recoverable strain capability comprises a nitinol material with an A_f greater than 37°C.
5. A method according to claim 3 wherein the outer surface material with limited recoverable strain capability comprises a material with lower resilience than the core material.
6. A method according to claim 3 wherein the outer surface material with limited recoverable strain capability comprises stainless steel.
7. A method according to claim 1, wherein the implantable medical device is subjected to surface modification prior to pre-straining.
8. A method according to claim 1 wherein the pre-straining results in at least about 0.30% nonrecoverable tensile strain.

9. A method according to claim 1 wherein the pre-straining results in at least about 0.40% nonrecoverable tensile strain.
10. A method according to claim 1 wherein the pre-straining results in at least about 0.50% nonrecoverable tensile strain.
11. A method according to claim 1 wherein the pre-straining results in at least about 0.80% nonrecoverable tensile strain.
12. A self-expanding endoprosthesis comprising a frame formed from nitinol exhibiting superelastic behavior at body temperature, the frame being configured to self-expand from a constrained delivery profile to an expanded operative profile, wherein the frame is pre-strained in a controlled manner to an extent that selectively induces at least about 0.20% non-recoverable tensile strain at or near the surface of at least one selected location on the frame during circumferential compaction of the frame to a smaller size.
13. A self-expanding endoprosthesis according to claim 12 wherein the pre-straining results in at least about 0.30% nonrecoverable tensile strain.
14. A self-expanding endoprosthesis according to claim 12 wherein the pre-straining results in at least about 0.40% nonrecoverable tensile strain.
15. A self-expanding endoprosthesis according to claim 12 wherein the pre-straining results in at least about 0.50% nonrecoverable tensile strain.
16. A self-expanding endoprosthesis according to claim 12 wherein the pre-straining results in at least about 0.80% nonrecoverable tensile strain.
17. An implantable article comprising nitinol metal alloy wire subjected to flexural fatigue test performed in a water bath at a temperature of $37\pm 1^{\circ}\text{C}$ and at a cyclic frequency of approximately 18 Hz, wherein a maximum principle mean tensile strain of 2.5% and an alternating strain of 0.3% are applied at an outside radius of an apex formed in the wire, said wire having a fatigue life of at least about 1,000,000 cycles prior to failure by fracture at the apex.

18. An implantable article according to claim 17 wherein said wire has a fatigue life of at least about 10,000,000 cycles prior to failure by fracture at the apex.
19. An implantable article according to claim 17 wherein said wire has a fatigue life of at least about 40,000,000 cycles prior to failure by fracture at the apex.
20. An implantable article according to claim 17 comprising a wire having a substantially round transverse cross section.
21. An implantable article according to claim 20 wherein said wire has a fatigue life of at least about 10,000,000 cycles prior to failure by fracture at the apex.
22. An implantable article according to claim 20 wherein said wire has a fatigue life of at least about 40,000,000 cycles prior to failure by fracture at the apex.
23. A method of forming a medical device comprising:
 - providing a computer program for use in finite element analysis, wherein the computer program calculates non-recoverable tensile strain;
 - employing the computer program to model the compressive residual strains for a medical device; and
 - designing the medical device so that non-recoverable tensile strain for the device is at least about 0.20% so as to improve the fatigue life of the medical device.
24. A method of forming a medical device comprising
 - providing a computer program for use in finite element analysis wherein the computer program models non-recoverable tensile strain for the medical device; and
 - designing the medical device so that the non-recoverable tensile strain of the medical device exceeds at least about 0.20% during the manufacture of the medical device.

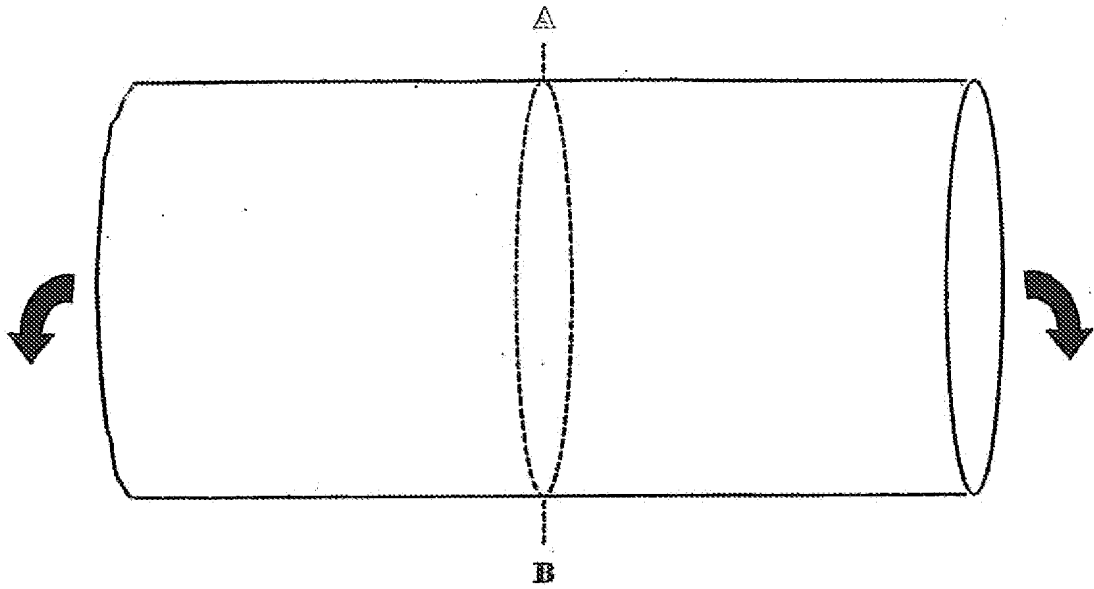


FIG. 1A

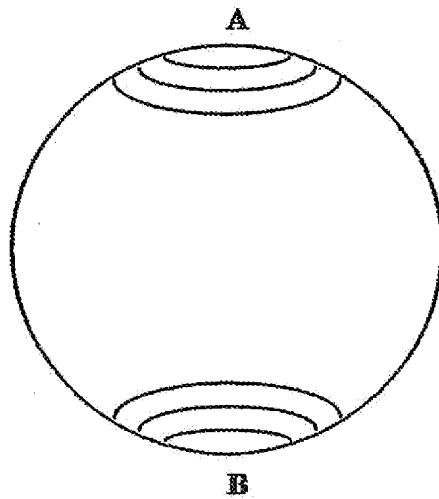


FIG. 1B

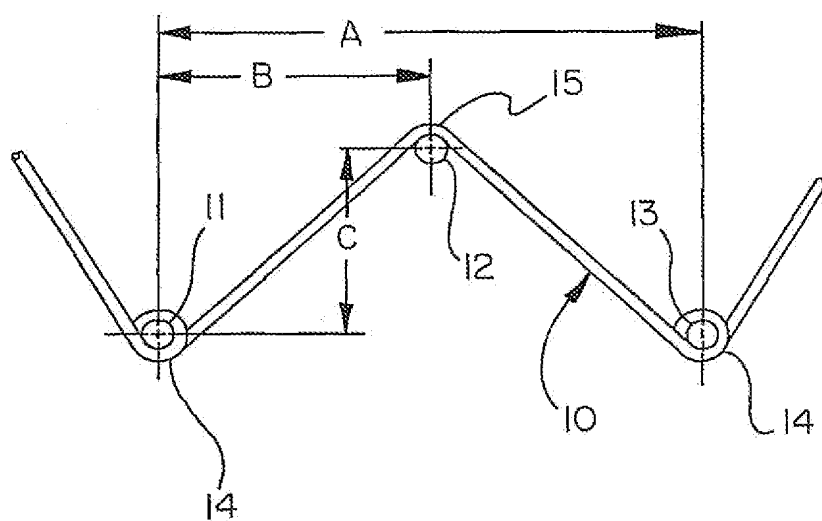
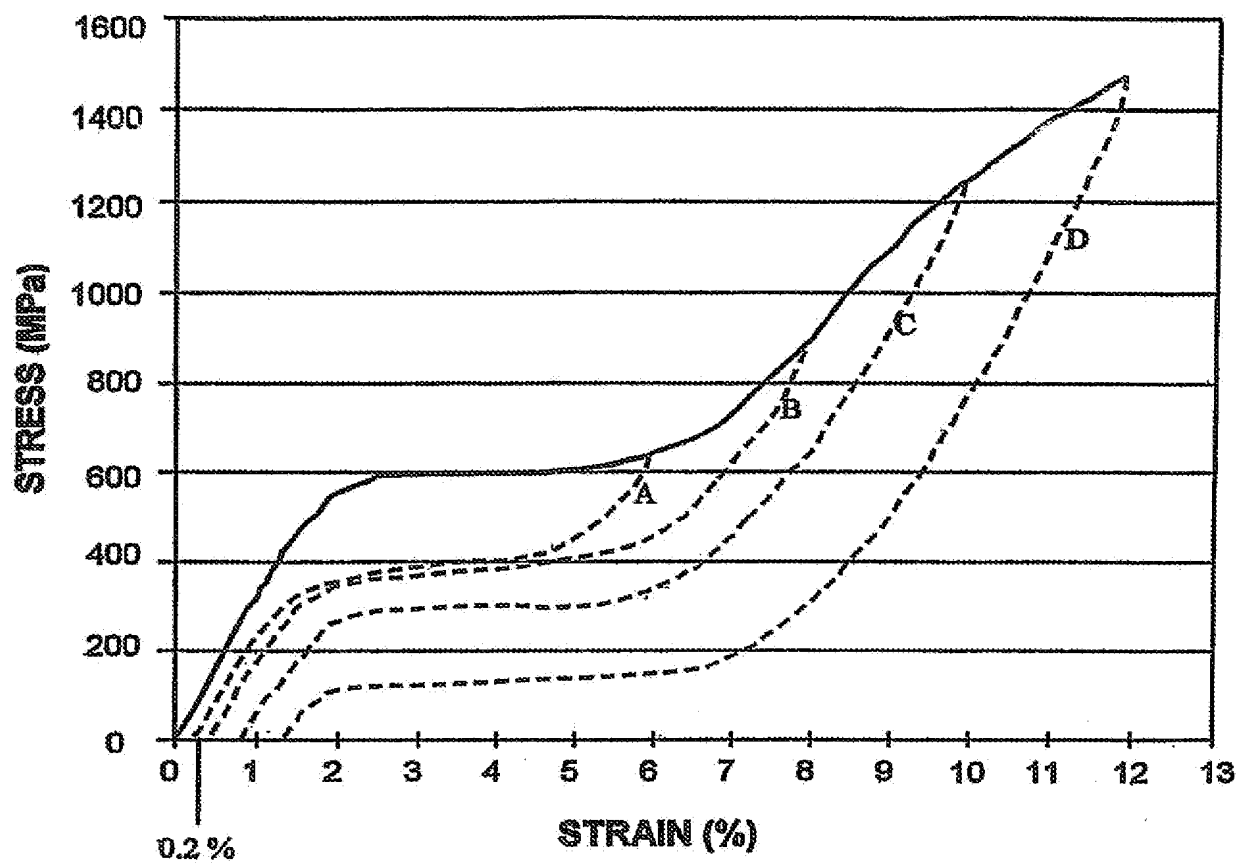
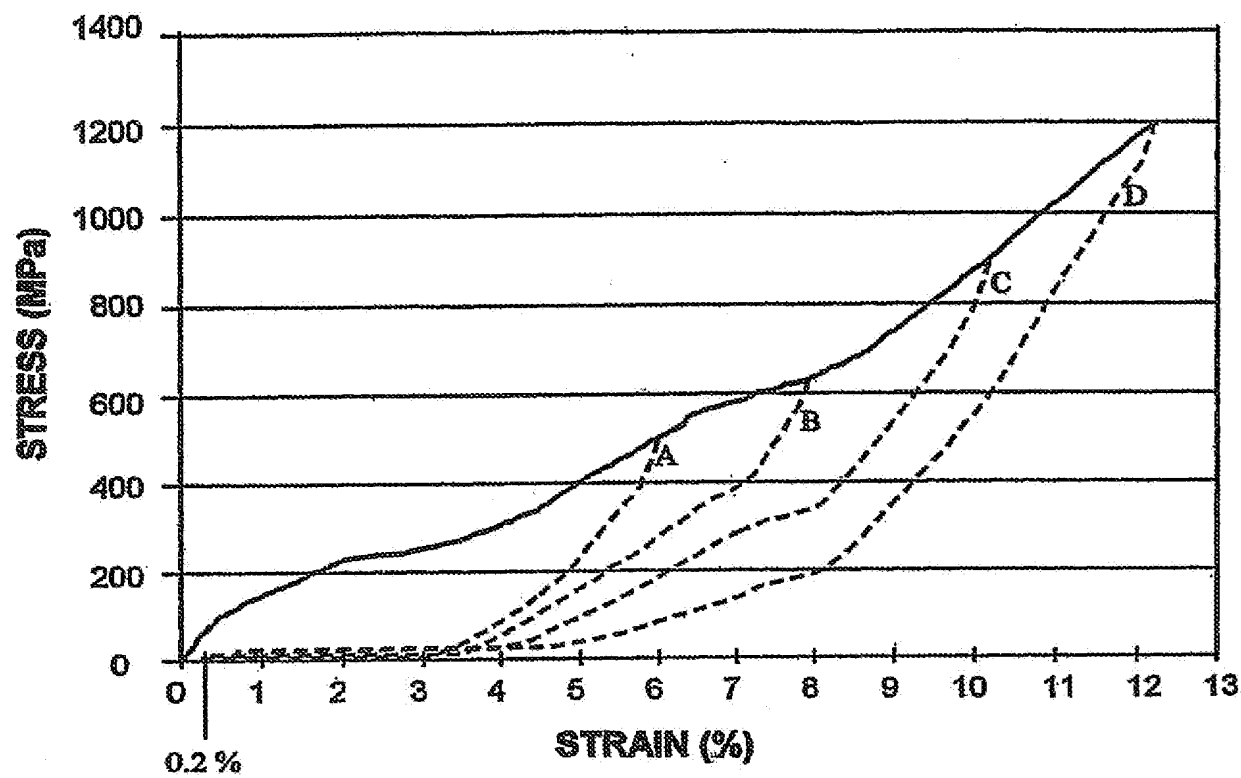
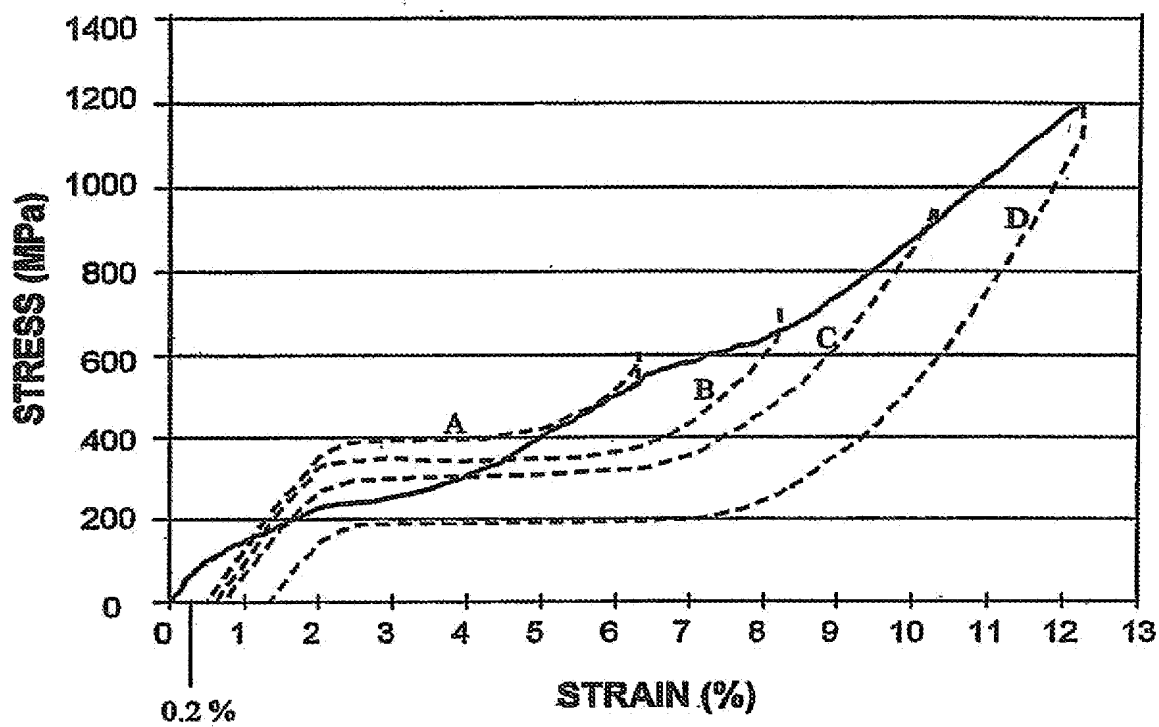
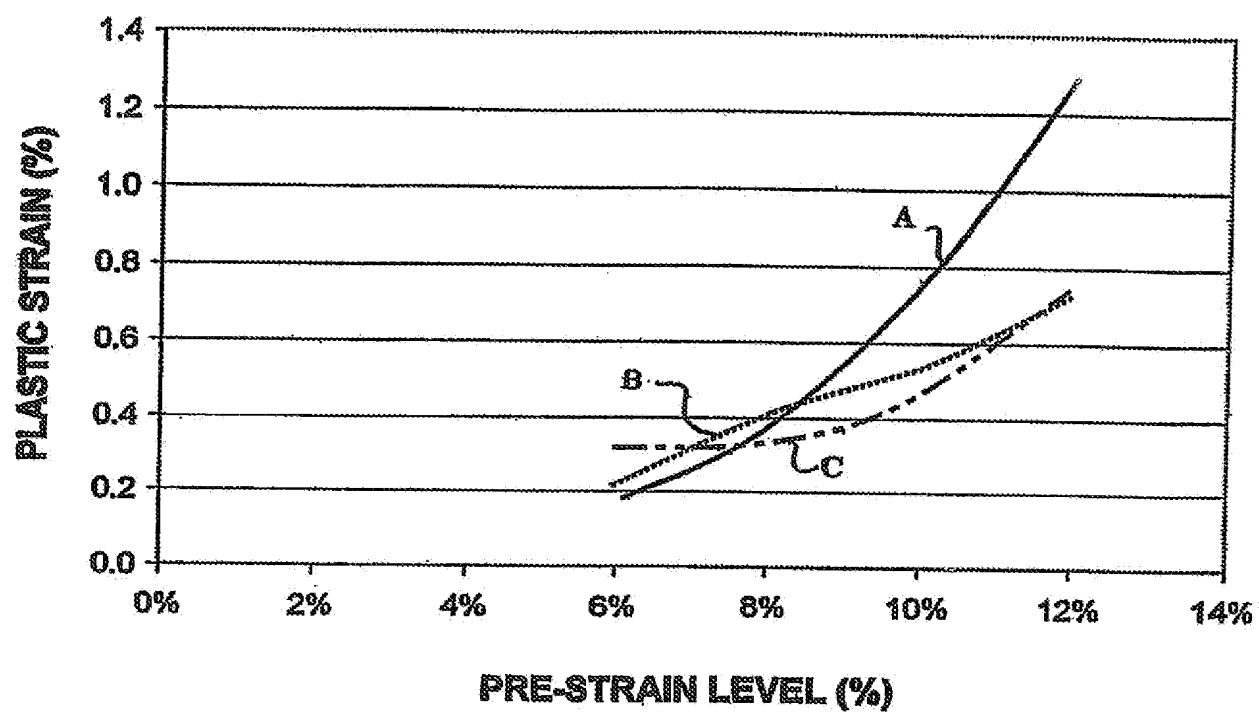


FIG. 1C

**FIG. 2**

**FIG. 3**

**FIG. 4**

**FIG. 5**

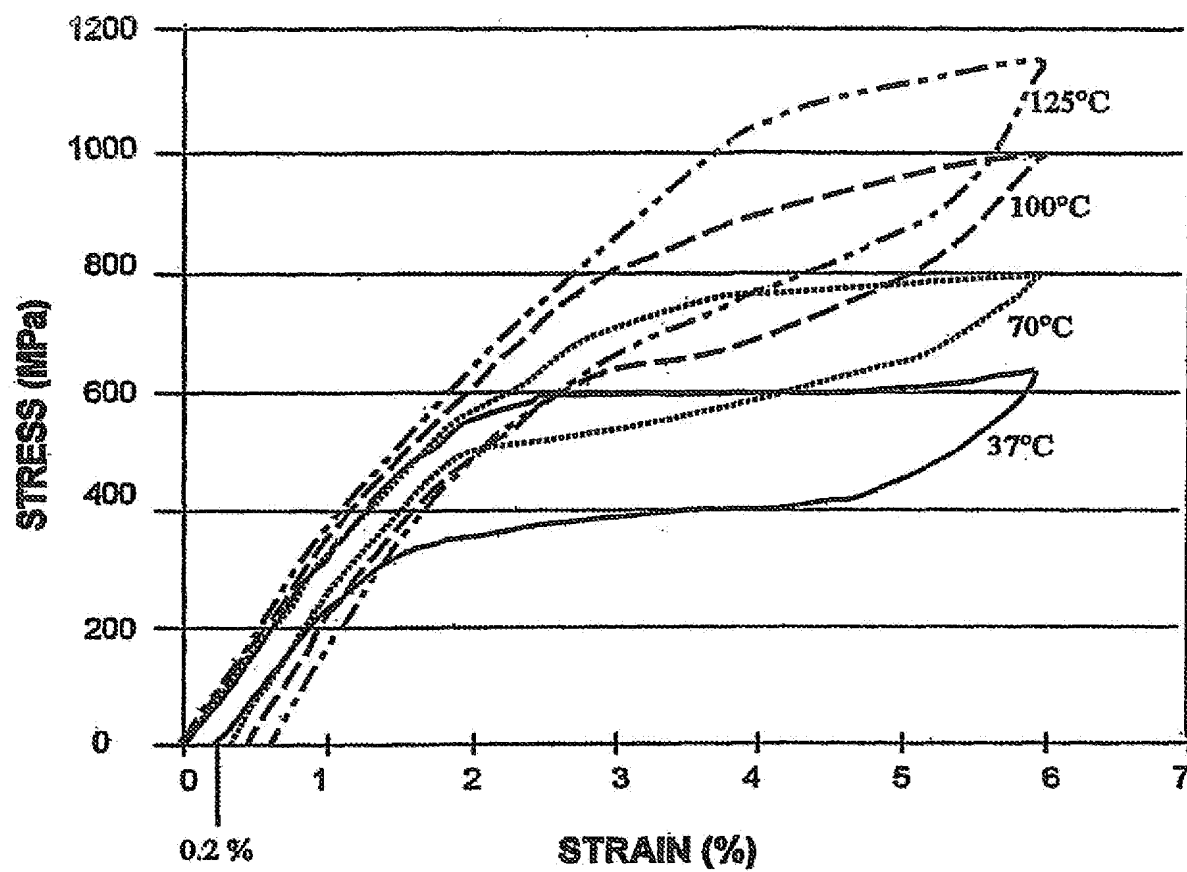
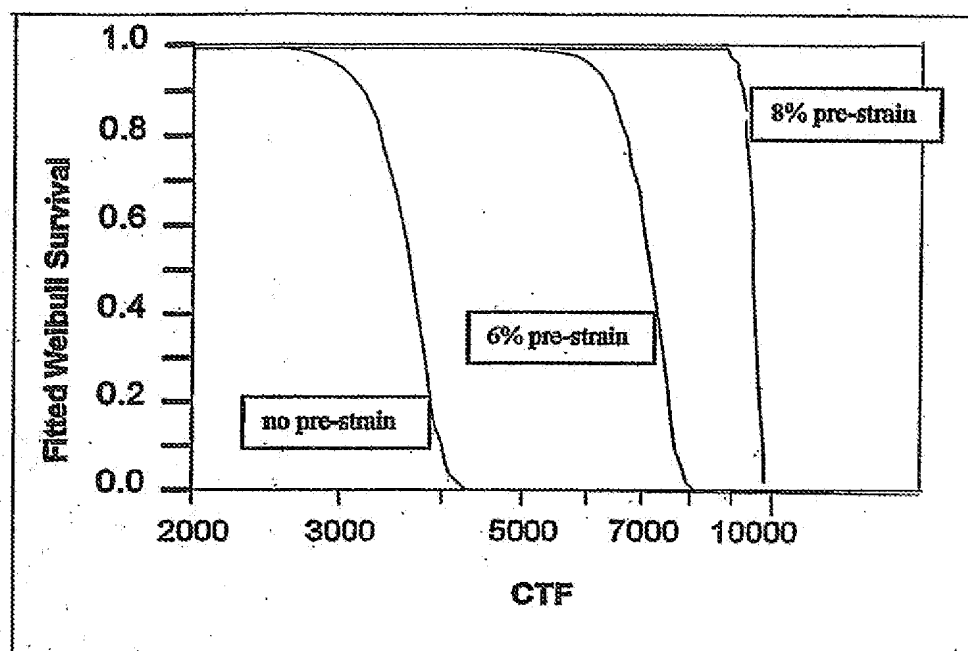
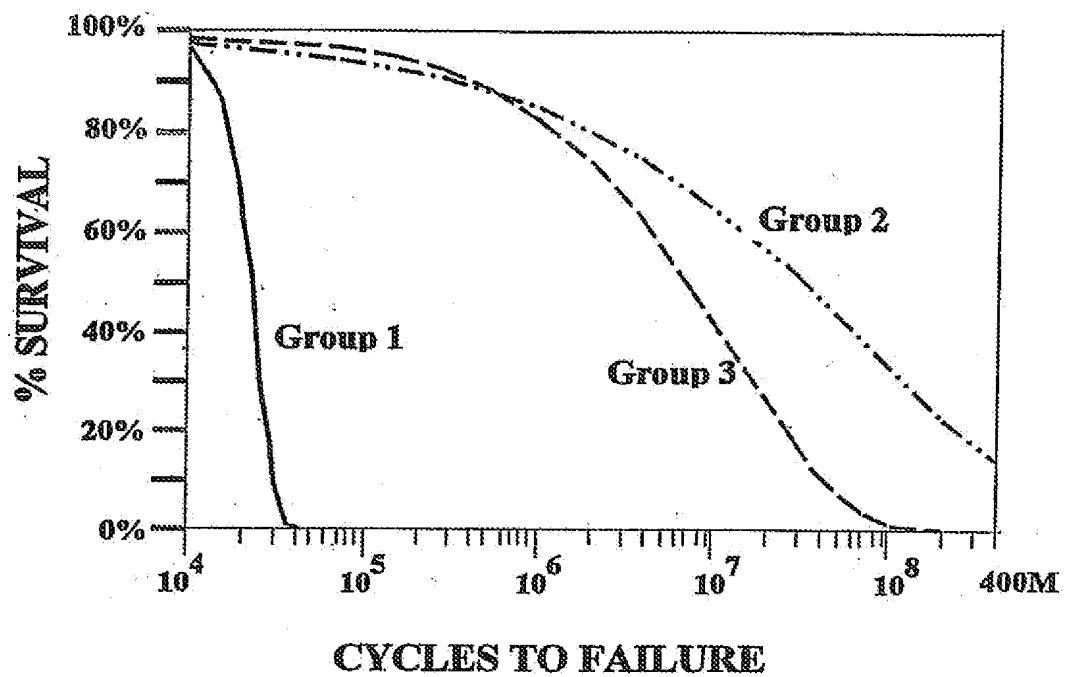


FIG. 6

**FIG. 7**

**FIG. 8**